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- 1. FIRST DAYS ACTIVITY BY TEAM INDICATES THE FOLLOWING:
- A. THE ENGINE WAS NOT ROTATING AT IMPACT. STOPPAGE WAS DUE TO TURBINE BLADE ENTANGLEMENT WHICH IS ATTRIBUTED AT THIS POINT TO FAILURE OF FOUR NBR ONE DISC CHRISTMAS THREE BLADE SECTIONS ALLOWING RELEASE OF 5 FIRST STAGE TURBINE BLADES. THERE IS NO EVIDENCE OF COMPRESSOR DAMAGE OTHER THAN THAT ASSOCIATED WITH IMPACT.
- B. PILOTS EJECTION APPEARS TO HAVE BEEN NORMAL EXCEPT THAT IT WAS ACCOMPLISHED AT A VERY LOW ALTITUDE. (APPROX 300 FT) THERE IS NO EVIDENCE THAT EJECTION WAS THROUGH CANOPY OR THAT PILOT COLLIDED WITH CANOPY AFTER EJECTION. EYE WITNESS WHO THOUGHT HE SAW PILOT IMPACT CANOPY WAS UNDOUBTEDLY MISLED DUE TO HIS POSITION WHICH WAS IN LINE WITH FLT PATH.

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THIS IS SECOND FAILURE OF A CONVERTED J-75-P-17 ENG IN THE IDEALIST PROGRAM AND COULD BE RELATED TO THE FIRST STAGE TURBINE BLADF SHROUD SHINGLING PROBLEM (LAST OCCURED WITH IN APRIL 1965) IT IS QUITE POSSIBLE

THIS TYPE FAILURE IS ASSOCIATED ONLY VITH THOSE FORMER

-17 ENGINES USED IN F-106 PROGRAM. RECOMMEND THAT ACFT

UNTIL FURTHER INVESTIGATION ESTABLISHES EXACT CAUSE OF

HAVING J-75-P-17 CONVERSION ENGINES INSTALLED BE GROUNDED

FAILURE. ALL OTHER ENGINES SHOULD BE INSPECTED FOR SHINGLING

AFTER EACH 15 HRS OF FLT VICE CURRENT 25 HR PROCEDURE UNTIL

DO NOT HAVE COMPLETION OF INVESTIGATION.

-17 CONVERSIONS INSTALLED.

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